(once amended) A turbine nozzle for a gas turbine engine, said nozzle comprising:

a pair of identical airfoil vanes coupled by at least one platform that is formed integrally with said airfoil vanes, each said airfoil vane comprising a first sidewall and a second sidewall connected at a leading edge and a trailing edge to define a cavity therebetween; and

at least one insert configured to be inserted within said airfoil vane cavity and comprising a first sidewall and a second sidewall, said insert first sidewall comprising a first plurality of openings extending therethrough for directing cooling air towards at least one of said airfoil vane first and second sidewalls, said insert second sidewall comprising a second plurality of openings extending therethrough for directing cooling air towards at least one of said airfoil vane first and second sidewalls, said first plurality of openings configured to facilitate more vane sidewall cooling than said second plurality of openings, said first plurality of cooling openings is greater than said insert second plurality of cooling openings.

(once amended) A nozzle in accordance with Claim wherein said pair of airfoil vanes further comprise a leading airfoil vane and a trailing airfoil vane, said at least one insert further comprises a first insert installed within said leading airfoil vane, and a non-identical second insert installed within said trailing airfoil vane, said inserts configured to facilitate cooling said trailing airfoil vane more than said leading airfoil vane.

## REMARKS

The Office Action mailed October 22, 2002 has been carefully reviewed and the foregoing remarks have been made in consequence thereof. A Submission of Marked Up Claims is submitted herewith.

Claims 1-5, 7-17, and 19 are pending in this application. Claims 1-3, 5, 13-17, and 19 stand rejected. Claims 7-12 are allowed. Claims 4, 6, and 18 are objected to.

The objection to Claim 17 is respectfully traversed. Specifically, Claim 17 has been amended, in accordance with the specification, to remove recitations to "upstream" and to "downstream". More specifically, Claim 17 has been amended to recite "a leading airfoil vane and a trailing airfoil vane." Accordingly, for at least the reasons set forth above, Applicants request the objection to Claim 17 be withdrawn.